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## FABRICATION AND TESTING OF HYBRID MATERIALS KEVLAR BASALT AND CHOPPED MATE WITH HORN POWDER

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### ABSTRACT

Combining two or more component materials with dissimilar physical or chemical characteristics results in a composite material. and which, inside the completed construction, continue to exist separately and independently at the microscopic or macroscopic levels. Stated differently, the elements work together to produce a single substance, but they do not disintegrate or melt into one another. Fibre materials are becoming more and more commonplace, and their market share is steadily increasing. Comparing composite fibres to conventional engineering materials reveals that they are less expensive, lightweight, and ecologically benign. This study's main objective was to use GOAT horn powder, basalt, chopped mate, and kevlar. Combinations of BASALT+10% GOAT HORN POWDER, CHOPPED MAT FIBER+10% GOAT HORN POWDER, KEVLAR FIBER+10% GOAT HORN POWDER, BASALT+ CHOPPED MAT FIBER+10% GOAT HORN POWDER, CHOPPED MAT FIBRE + KEVLAR FIBER+10% GOAT HORN POWDER, KEVLAR FIBRE+ BASALT FIBER+10% GOAT HORN POWDER, BASALT FIBRE + CHOPPED MAT FIBRE + KEVLAR FIBRE + 10% GOAT HORN POWDER will be provided by using the three reinforcements with epoxy resin previously mentioned. A typical addition to all compositions is 10% GOAT horn powder. Tensile, flexural, impact, and hardness tests were used to ascertain the qualities of these composites, which were then manufactured into crossover materials that are evaluated provisionally in accordance with ASTM standards. Consider using a real-time application automobile bumper after determining which material is ideal. Determine the stress, strain, deformation, and shear stress using the best material and Gmat material by creating in Catia programme and doing the static analysis in Ansys software.

## INTRODUCTION OF FIBERS

According to most definitions, a composite is a blend of at least two distinct components. Fiber-reinforced polymer (FRP) is a popular composite in which the polymer serves as a lattice and the strands serve as support. The structure functions as glue, holding the strands together and increasing their weather resistance. Because the grid, unlike the network material, is more fragile than the filaments, mechanical properties such as hardness, strength, and sturdiness are required to let them to expand when they are united. Because filaments are often orientated in the same direction (unidirectional), their properties are anisotropic, resulting in outstanding properties along the fiber course. Near isotropic properties can be produced if the strands are put freely in a multidimensional framework (multidirectional). Polymer lattice composites (PMCs) are a form of composite made of polymer lattices (polymer grid composite). Composite matrix composites (CMCs) and metal network composites (MMCs) are two forms of composites (clay framework composites). Because composites are equally robust but much lighter than metals, they are routinely utilized to replace them. This idea, in any case, only applies to PMCs.

## INDUSTRIAL APPLICATIONS OF FIBER REINFORCED COMPOSITES MATERIALS

### (a) Military and aerospace applications



**Figure 12 Light weight natural composites military Helmet**



**Figure 13 Carbon Fiber Reinforced Composite Surfboards**

## LITERATURE REVIEW

[1] The motivation behind this exploration is to set up cut evidence material comprised of shear thickening liquid (STF) and Kevlar fiber. In this exploration, silica/ethylene glycol suspension was ready for the utilization as STF and it was assessed by remoter. From the outcomes, it was seen that STF displayed a converse fluid strong progress at a specific shear. Kevlar was treated as STF by 1 plunge 1 nip technique and mechanical and cut safe properties were examined. Subsequent to survey both the outcomes, STF impregnation exhaustively overhauled the wound opposition of Kevlar against the spike dangers and the wellbeing part of Kevlar was additionally expanded extensively.

[2] The cut safe covering was performed by considering one STF, smoldered silica/ethylene glycol suspension of Kevlar texture to upgrade the presentation of the material. From this exploration, broad redesigns in cut insubordination were seen particularly in extreme speed stacking condition. It was seen that the expansion didn't change or decay the adaptability of STF. From the outcomes, we caused that seethed silica/Kevlar composite texture would be a fine material for body defensively covered applications

[3] Moreover, various materials that are consolidated to deliver an individual part. The two materials that were explored were Kevlar poly (p-phenylene terephthalamide) and santoprene. Here in this examination, the Kevlar fiber was utilized two,

the principal type was, it was utilized without changing it and the subsequent kind was utilized in the wake of adjusting it.

[4] The Kevlar which was utilized without altering reinforced the santoprene to cite a degree and it overhauled not many properties of the composite, in particular low strain modulus and rigidity however it likewise had a disadvantage, extending at break decreased intensely. To conquer this, the Kevlar was adjusted and henceforth its surface was hydrolyzed maleic anhydride-grafted polypropylene (MA-g-PP). There were clear benefits of utilizing the changed Kevlar over the stock one. The properties upgraded and the downsides were diminished to nil. This blend showed further developed pressure circulation because of better surface holding between the fiber and lattice .

[5] From that point, Fluorinated and Oxy fluorinated Short Kevlar Fiber-Reinforced Ethylene Propylene Polymer This paper looks at crude Kevlar and surface treated Kevlar. Assessment on its warm properties showed an expansion in warm dependability and capacity modulus because of steady support of filaments. It is additionally noticed that it keeps on expanding on account of bull fluorinated and fluorinated Kevlar fiber-built up EP.

[6] Fluorinated and oxy fluorinated Kevlar fiber's rigidity expanded significantly. This presumes that fluorination and oxy fluorination affected the surface morphology giving better bond of filaments and the framework

[7] Moreover, correlation of properties like glasslike, warm, mechanical of syndiotactic polystyrene composites with surface adjusted Kevlar fiber.

[8] In contrast with ox fluorinated Kevlar built up fiber, fluorinated Kevlar fiber arrives at higher crystallization temperature. There is a huge expansion in warm conductivity in the event of changed Kevlar fiber support. It additionally moves to a higher worth displayed by differential examining calorimeter and dynamic mechanical examination. A solid bond between sPS lattice and oxy-fluorinated Kevlar fiber was discovered and it

is by all accounts better compared to different composites displayed by nuclear free microscopy. From this exploration it was discovered that it helps up the warm dependability and capacity modulus of the composite

[9]The cut safe covering was performed by considering one STF, smoldered silica/ethylene glycol suspension of Kevlar texture to upgrade the presentation of the material. From this exploration, broad redesigns in cut insubordination were seen particularly in extreme speed stacking condition. It was seen that the expansion didn't change or decay the adaptability of STF. From the outcomes, we caused that seethed silica/Kevlar composite texture would be a fine material for body defensively covered applications

[10] Moreover, various materials that are consolidated to deliver an individual part. The two materials that were explored were Kevlar poly (p-phenylene terephthalamide) and santoprene. Here in this examination, the Kevlar fiber was utilized too, the principal type was, it was utilized without changing it and the subsequent kind was utilized in the wake of adjusting it.

[11] The Kevlar which was utilized without altering reinforced the santoprene to cite a degree and it overhauled not many properties of the composite, in particular low strain modulus and rigidity however it likewise had a disadvantage, extending at break decreased intensely. To conquer this, the Kevlar was adjusted and henceforth its surface was hydrolyzed maleic anhydride-grafted polypropylene (MA-g-PP). There were clear benefits of utilizing the changed Kevlar over the stock one. The properties upgraded and the downsides were diminished to nil. This blend showed further developed pressure circulation because of better surface holding between the fiber and lattice.

[12] However, the properties of Kevlar polypropylene dependent on composite material under high strain rate stacking utilizing Split Hopkinson Pressure bar (SHPB). Level covered Kevlar composite of 16, 24 and 30 layer where examination formed and laser machined to acquire round and hollow example of wanted shape

dependent on SHPB try. To report compressive material conduct as extension of developing strain rate, the pressure strain plots were get and broke down. The examinations propose that for better execution of composite

[13] In this work creators are accounted for about the execution and analyze the qualities of the novel auxetic Kevlar composite. This examination was especially founded on break and effect qualities. To investigations and separate, Kevlar interlinked composite was utilized alongside polyurethane examination and without it. Short nylon filaments of two recognized fiber estimation and 3 recognized fibers recognized fiber densities were combined.

### FABRICATIONS AND EXPERIMENTAL SETUPS

#### OBJECTIVES:

Following are the objectives that have been outlined:

1. Fabrication of a new class of epoxy-based hybrid composite reinforced Basalt with 10% Horn powder, Chopped mate with 10% Horn powder, Kevlar with 10% Horn powder, Kevlar Basalt with 10% Horn powder, Chopped mate Basalt with 10% Horn powder, Chopped mate+ Kevlar with 10% Horn powder, Basalt +Chopped mate + Kevlar with 10% Horn powder.
2. Investigation and Evaluation of Mechanical properties such as tensile strength, flexural strength, hardness.
3. To study the potential utilization of Basalt with 10% Horn powder , Chopped mate with 10% Horn powder, Kevlar with 10% Horn powder, Kevlar Basalt with 10% Horn powder, Chopped mate Basalt with 10% Horn powder, Chopped mate+ Kevlar with 10% Horn powder, Basalt +Chopped mate Kevlar with 10% Horn powder as reinforcement material in epoxy-based composites for various applications.
4. Take out and cool the specimen until room temp about 24 hrs and Ensure proper weighing is maintain.

5. To assess whether the fabricated hybrid composite can be used as an alternate material
6. For synthetic fiber reinforced composites.
7. Cut to appropriate dimension as per ASTM Standard and Impact, flexural, hardness and tensile
8. Behavior of fabricated were calculated by various Mechanical testing's done.
9. To design and Analysis of Car Bumper using Existing dimension.
10. Increase the Car bumper strength compared to the existing materials we consider.
11. The Kevlar Material with New Material who is the best properties find out in 7 orientations.
12. Find out the Von-misses stresses, Shear stress and deformations in static analysis.
13. Finally concluded the suitable material for the Car Bumper .

#### METHODOLOGY

I: Collecting data and information identified with regular filaments and manufactured strands.

II: Arrangement of examples utilizing hand layup procedure

III: Conducting tensile, impact, hardness and flexural tests.

IV: Plot outlines for the outcomes and manual estimations are directed.

V: Identify better fiber among 7 Fiber.

VI:A completely parametric model of the Car bumper is made in CATIA software.

VII: Model got in IGS. Broke down utilizing ANSYS 15.0(workbench), to acquire stress, strain and deformations.

VIII: Taking limit conditions and conducting static examination.

IX: Finally, we contrast better among these seven outcomes acquired from ANSYS and looked at changed materials.

## MATERIALS

On among various kinds of pitches and hardener. Epoxy LY556, hardener HY951 and Horn powder in all orientations is picked. The materials taken to manufacture the examples are Kevlar, basalt and chopped matt. These are taken in the various proportions and various mixes. The seven distinct composites are examined the tensile, impact, hardness and flexural tests.

### HORN POWDER:



Figure 14 Horn powder

| Horn powder                 | Values  |
|-----------------------------|---------|
| Density(g/cm <sup>3</sup> ) | 1.56    |
| Tensile Strength(Mpa)       | 540     |
| Young's Modulus (Gpa)       | 200     |
| Melting Point               | 1566 °C |
| Possion's Ratio(u)          | 0.29    |

Table 2 Material properties of Horn powder

### KEVLAR

Kevlar has numerous applications, going from bike tires and hustling sails to tactical armor carriers, all because of its high elasticity to-weight proportion; by this action it is multiple times more grounded than steel. It likewise is utilized to make present day walking drumheads that withstand high effect. Kevlar is a manufactured plastic, and it's made of a chemical compound called **poly-para-phenylene terephthalamide**. This chemical is made from creating a chemical reaction between an acid and a chemical solution containing nitrogen and hydrogen

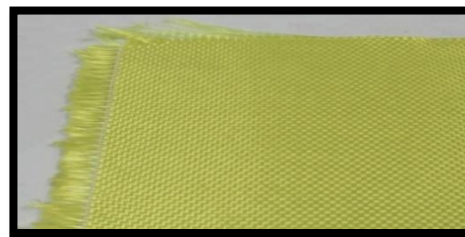


Figure 15 Kevlar

### BASALT FIBER

The basalt is simply washed and then melted. The manufacture of basalt fiber requires the melting of the crushed and washed basalt rock at about 1,500 °C (2,730 °F). The molten rock is then extruded through small nozzles to produce continuous filaments of basalt fiber. A hard, dense volcanic rock that can be found in most countries across the globe, basalt is an igneous rock, which means it began in a molten state. For many years, basalt has been used in casting processes to make tiles and slabs for architectural applications. Additionally, cast basalt liners for steel tubing exhibit very high Abrasion resistance in industrial applications. In crushed form, basalt also finds use as aggregate in concrete.

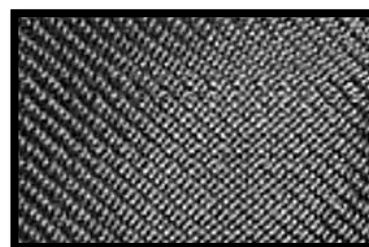


Figure 16 Basalt

### CHOPPED MATT

Chopped strand mat is a non woven reinforcement fiber for the production of fiber-reinforced plastic.



Figure 17 Chopped matt

### FABRICATION OF COMPOSITE SPECIMENS (HAND LAYUP)

Hand lay-up procedure is the straightforward and least expensive strategy for composite handling.

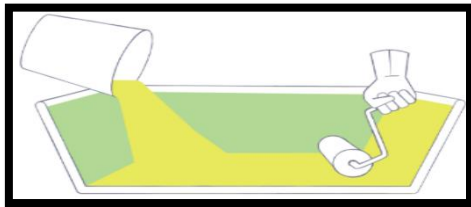


Figure 18 Fabrications of composite specimens

With the help of a brush, the polymer is evenly distributed. Then second layer of fiber is placed on the polymer surface and another layer of polymer is applied after this is closed with another thin plastic sheet after squeezer is moved with a gentle pressure on the thin plastic sheet to remove air. The consequential mold is cured for 24 hours at room temperature.



Figure 19 Complete sequential process for fabrication



Figure 20 Tensile test specimens

### FLEXURAL TESTING OF COMPOSITES

Three point bowing test are carried out as per ASTM-D790M-86 test procedure 1, system A to extract flexural properties, the specimens are 100 mm long, 25 mm wide and 4.5 mm thick. Two indistinguishable specimens are subjected for flexural testing. In three point bowing test, the external rollers are 70 mm separated and specimens are subjected at a strain rate of 0.2 mm/min. Flexural stress are determined by the following relations.

$$\text{Flexural stress } S = \frac{3 P L}{2 b t^2}$$

P= load in N

L= length between supports (70mm)

b= Width in mm

d= Thickness in mm



Figure 21 specimen's flexural test

### IMPACT TESTING OF COMPOSITES

Impact test is also known as Charpy V notch, Impact tester was Charpy analyzer supplied by M/S International Equipments, Mumbai, was used to test the impact properties of fiber Reinforced composite specimen. The Impact tester has four working abilities of effect quality i.e. 0-2.71 J, 0-5.42 J, 0-10.84 J and 0-21.68 J, with a base determination on every size of 0.02J, 0.05 J, 0.1 J

and 0.2 J individually .Four scales and comparing mallets (R1,R2,R3,R4) are presented in equipment.

Standard test procedure, ASTM D256-97, for effect properties of fiber composites has been used to examine the unidirectional composite specimens. The specimens to be examined are of dimensions 63.5mm long, 12.36mm wide and 6mm in thick. A V-point is placed in impact tester record having an included point of 450 at the focal point of the specimen, and at 90° to the specimen pivot. The profundity of the specimen to be examined under the indent is 2 mm.



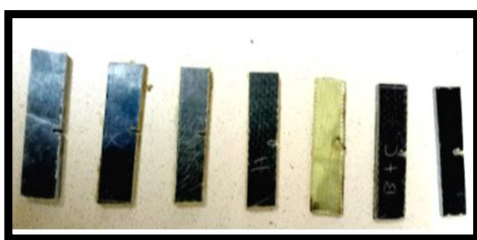
**Figure 22 Impact machine for impact testing**

Impact strength was calculated by the following relation

$$\sigma = \frac{2P}{A}$$

P= Energy observed in J

A= Area in mm



**Figure 23 before testing impact testing**

In this test, the configuration is limited to unidirectional and continuous fibers equal to the length of the specimen. The hardness properties of the composites are studied by applying indentation load normal to fibers diameter and normal to fiber length. The effect of fiber loading and post curing

time on Rockwell hardness is illustrated in Figures 3 and 4. Generally, fibers that increase the module of composites increase the hardness of the composite. This is because hardness is a function of the relative fiber volume and modulus.



**Figure 24 before Testings hardness test**

**TESTING IMAGES AT LABORATORY:**



**Figure 25 TESTING PROCESS OF SPECIMENS**

**AFTER TESTING:**





**Figure 26 AFTER TESTING FLEXURAL SPECIMENS**



**Figure 27 AFTER TESTING OF IMPACT SPECIMENS**



**Figure 28 AFTER TESTING OF HARDNESS SPECIMENS**

**RESULTS AND DISCUSSION**

**MECHANICAL CHARACTERISTICS OF COMPOSITES**

The properties of the Kevlar, Basalt, chopped mate, Kevlar + basalt, basalt+chopped matt, Kevlar +chopped matt & Kevlar basalt+ chopped matt fibers and basalt reinforced epoxy hybrid composites with of fiber under this investigation are presented in below Table 4.1. I have taken each composite for each test. Details of processing of these composites and the tests conducted on them have been described in the previous chapter. The mechanical properties of Synthetic fiber reinforced composites are largely

depends on the chemical, structural composition, fiber type and soil conditions and also on atmospheric conditions at the time of fabrication of the specimens.

| S.NO | Composite                     | Tensile test |                  | Flexural test |                  | Impact test Strength in J |
|------|-------------------------------|--------------|------------------|---------------|------------------|---------------------------|
|      |                               | Load in N    | Elongation in mm | Load in N     | Elongation in mm |                           |
| 1    | Basalt                        | 6900         | 9.2              | 1960          | 8.5              | 3.7                       |
| 2    | Kevlar                        | 9495         | 21.3             | 1450          | 7.0              | 5.45                      |
| 3    | Chopped matt                  | 8920         | 17.1             | 1400          | 8.55             | 5.4                       |
| 4    | Basalt+ Kevlar                | 8620         | 10.6             | 1550          | 7.3              | 5.3                       |
| 5    | Basalt+ Chopped matt          | 7125         | 8.4              | 1810          | 8.0              | 3.9                       |
| 6    | Kevlar+ Chopped matt          | 9750         | 23.2             | 1070          | 8.6              | 5.7                       |
| 7    | Kevlar + Basalt+ Chopped matt | 9525         | 22.1             | 1923          | 6.8              | 5.5                       |

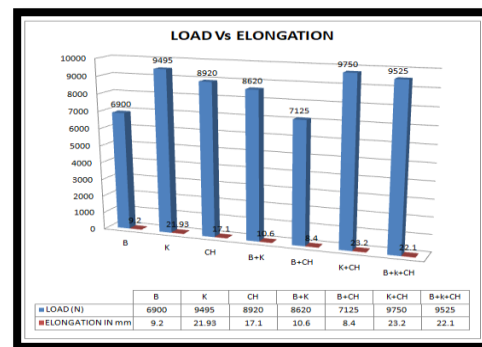
**Table: 1 Specimen testing results TENSILE STRENGTH**

Fabrication and testing successfully completed in this project the tensile properties of Kevlar, Basalt, chopped mate, Basalt + chopped matt, Kevlar + Basalt, Kevlar+ Basalt+chopped matt, Kevlar+ chopped matt fabricated by using hand lay-up method. The tensile strength was calculated by the relation

| S.NO | Composite                    | Load in N | Elongation In mm | Tensile strength N/mm <sup>2</sup> | % of elongation |
|------|------------------------------|-----------|------------------|------------------------------------|-----------------|
| 1    | Basalt                       | 6900      | 9.2              | 9.50                               | 5.60            |
| 2    | Kevlar                       | 9495      | 21.93            | 13.07                              | 12.98           |
| 3    | Chopped mate                 | 8920      | 17.1             | 12.28                              | 10.42           |
| 4    | Basalt+ Kevlar               | 8620      | 10.6             | 11.87                              | 6.46            |
| 5    | Basalt+ Chopped mate         | 7125      | 8.4              | 9.81                               | 5.12            |
| 6    | Kevlar+ Chopped mate         | 9750      | 23.2             | 13.42                              | 14.14           |
| 7    | Basalt+ Kevlar+ Chopped mate | 9525      | 22.1             | 13.11                              | 13.47           |

**Table: 2 tensile test results for 7 composites**

After successful completion of the tensile strength we are getting maximum values for the Kevlar with chopped mate 9750 N.



**Graph 1 tensile test result graph**

### FLEXURAL STRENGTH

Fabrication and testing successfully completed in this project the flexural strength of basalt, Kevlar, chopped mate and basalt chopped mate, basalt Kevlar, Kevlar chopped mate, basalt Kevlar chopped mate with Horn powder are fabricated by using hand lay-up method. The flexural strength was calculated based the following relation

$$\text{strength } S = \frac{3 P L}{2 b t^2}$$

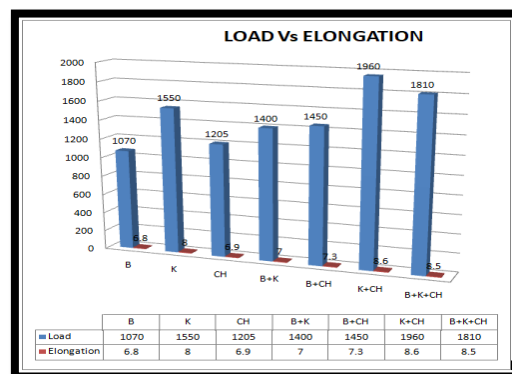
P= load in N; L= length between supports (70mm);b= Width in mm; d= Thickness in mm

After testing the Flexural strength and elongation are summarized in table. The percentages of elongations for all the composites are also calculated with the following formula.

$$\% \text{ elongation} = \frac{\text{change in length}}{\text{original length}} \times 100$$

- ❖ Basalt : % of elongation =  $\frac{6.8}{70} \times 100 = 9.71\%$
- ❖ Kevlar: % of elongation =  $\frac{8}{70} \times 100 = 11.42\%$
- ❖ Chopped matt: % of elongation =  $\frac{6.9}{70} \times 100 = 9.85\%$
- ❖ Basalt+Kevlar: % of elongation =  $\frac{7}{70} \times 100 = 10\%$
- ❖ Basalt+chopped matt: % of elongation =  $\frac{7.3}{70} \times 100 = 10.42\%$
- ❖ Kevlar + Chopped matt: % of elongation =  $\frac{8.6}{70} \times 100 = 12.28\%$
- ❖ Kevlar+ Basalt+Chopped matt : % of elongation =  $\frac{8.5}{70} \times 100 = 12.14\%$

Based on the flexural strength finally concluded that Kevlar+ Chopped mate of Horn powder epoxy composite possess high flexural strength compared to remaining composite as shown in figure. And Basalt+ Kevlar+ Chopped mate having a second highest flexural strength compared to remaining composite



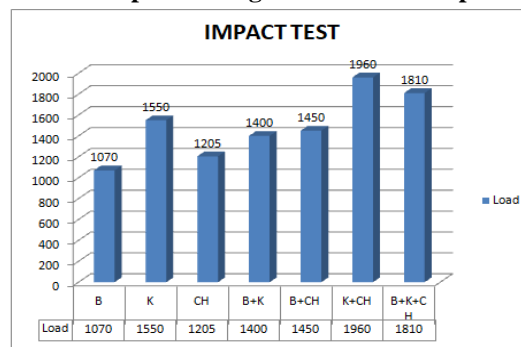
Graph 2 Flexural test result graph

### IMPACT STRENGTH

Fabrication and testing successfully completed in this project I also focused on impact strength of basalt, Kevlar, chopped mate and basalt chopped mate, basalt Kevlar, Kevlar chopped mate, basalt Kevlar chopped mate with Horn powder fabricated by using hand lay-up method. And finally concluded the Kevlar+ Chopped mate material possess high impact strength compared to remaining compositions as shown figure

| IMPACT TEST                  | JOULES |
|------------------------------|--------|
| Basalt                       | 3.7    |
| Kevlar                       | 5.45   |
| Chopped mate                 | 5.4    |
| Basalt+ Kevlar               | 5.3    |
| Basalt+ Chopped mate         | 3.9    |
| Kevlar+ Chopped mate         | 5.7    |
| Basalt+ Kevlar+ Chopped mate | 5.5    |

Table: 3 Impact testing results for 7 composites



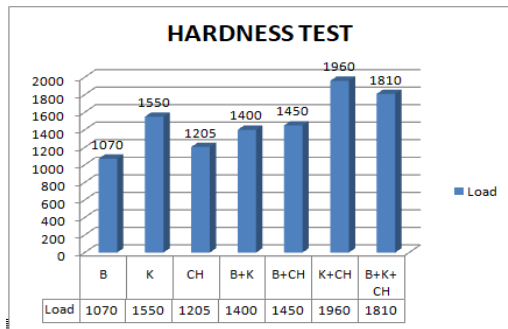
Graph 3 Impact strength result graph

**HARDNESS NUMBER:**

Brinell hardness values of these natural composites. Experiment gives the Kevlar with chopped mate having maximum Brinell hardness value 18.3 ,where Wt% ratio of resin & hardener: On the other hand, epoxy with basalt reveals the minimum hardness value 13.9.Brinell hardness vs. experiment number graph of the composite. Figure reveals the graph indicating Brinell hardness values corresponding to the experiment number. The graph shows, experiment with kevlar with chopped mate gives the higher value of Brinell hardness. On the other hand, experiment with basalt gives the lower Brinell hardness value.

| COMPOSITE                    | HARDNESS NUMBER |
|------------------------------|-----------------|
| Basalt                       | 13.9            |
| Kevlar                       | 16.5            |
| Chopped mate                 | 15.3            |
| Basalt+ Kevlar               | 14.3            |
| Basalt+ Chopped mate         | 14.1            |
| Kevlar+ Chopped mate         | 18.3            |
| Basalt+ Kevlar+ Chopped mate | 16.5            |

**Table: 4 Hardness testing results for 7 composites**



**Graph 4 Hardness number result graph**

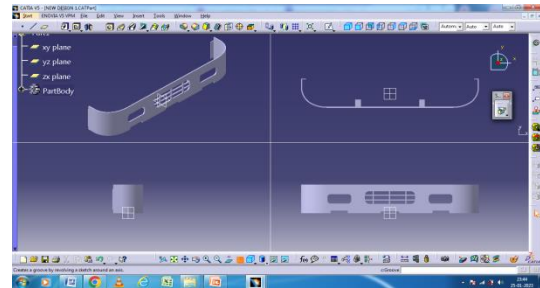
**INTRODUCTION TO CATIA**

Welcome to **CATIA (Computer Aided Three Dimensional Interactive Application)**. As a new user of this software package, you will join hands with thousands of users of this high-end CAD/CAM/CAE tool worldwide. If you are already familiar with the previous releases, you can upgrade your designing skills with the tremendous improvement in this latest release.

**.DESIGN PROCEDURE IN CATIA:**

Design process of maruthi suzuki a lot car bumper Go to the sketcher workbench create the 1200x300

c shape using profile and thickness is 8mm after apply pad using part design workbench again go to the front view xy plane create the front part gill area and light area apply pocket as per dimensions .



**Figure 29 Multiple view of car bumper**

**INTRODUCTION TO ANSYS**

ANSYS is a large-scale multipurpose finite element program developed and maintained by ANSYS Inc. to analyze a wide spectrum of problems encountered in engineering mechanics.

**PROGRAM ORGANIZATION:**

The ANSYS program is organized into two basic levels:

- Begin level
- Processor (or Routine) level

The Begin level acts as a gateway into and out of the ANSYS program. It is also used for certain global program controls such as changing the job name, clearing (zeroing out) the database, and copying binary files. When you first enter the program, you are at the Begin level.

**FINITE ELEMENT METHOD:**

The Basic concept in FEA is that the body or structure may be divided into smaller elements of finite dimensions called “Finite Elements”. The original body or the structure is then considered as an assemblage of these elements connected at a finite number of joints called “Nodes” or “Nodal Points”. Simple functions are chosen to approximate the displacements over each finite element. Such assumed functions are called “shape functions”. This will represent the displacement with in the element in terms of the displacement at the nodes of the element.

**Basic Steps in FEA**

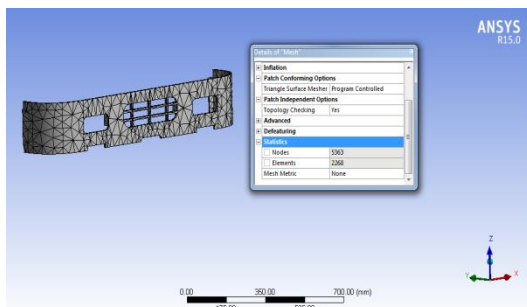
- Discretization of the domain
- Application of Boundary conditions
- Assembling the system equations
- Solution for system equations
- Post processing the results.

**MATERIAL PROPERTIES:**

| MATERIALS PROPERTIES                | KEVLAR             | KEVLAR+CHOPPEDMATE +10grms horn powder |
|-------------------------------------|--------------------|--|
| Density(kg/m <sup>3</sup> )         | 1380               | 1423                                   |
| Poisson's ratio                     | 0.35               | 0.32                                   |
| Young's Modulus(GPa)                | 76                 | 84                                     |
| Tensile Strength(N/m <sup>2</sup> ) | 29*10 <sup>6</sup> | 31.5*10 <sup>6</sup>                   |

**MESH AND BOUNDARY CONDITIONS:**

The meshed model of bullet proof jacket of nodes =10129 and elements is 4845 Using Material as finalized material from above mentioned Kevlar with chopped matt call it as a hybrid and existing material for car bumper of alto Maruthi car

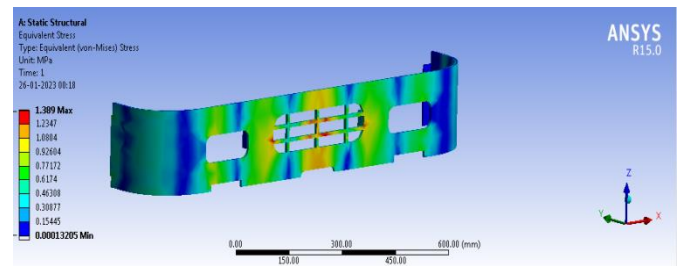


**Figure 30 MESHING: NODES: 5363  
ELEMENTS: 2268**

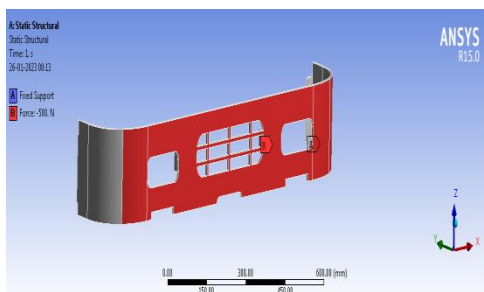
A static analysis calculates the effects of study loading conditions on a structure, while ignoring inertia and damping effects, such as those caused by time varying loads. A static analysis can however include steady inertia loads and time varying loads that can be approximated as static equivalent loads. Static analysis is used to determine the displacements, stresses, strains and forces in structures or components caused by loads that do not induce significant inertia and damping effects. Steady loading and response conditions are assumed, i.e. the loads and the structure's responses are assumed to vary slowly with respect to time. The kinds of loading that can be applied in static analysis include:

- Externally applied forces and pressures.
- Steady state inertial forces
- Imposed displacement
- Temperatures

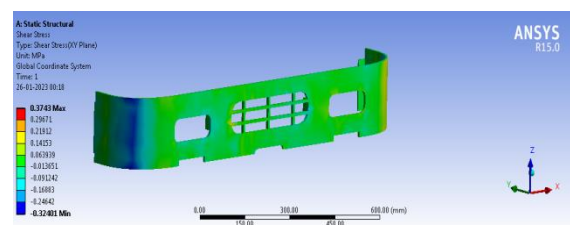
**Static Analysis of Kevlar fiber + 10grms Horn powder Material:**



**Figure 32 Von-misses stress of Kevlar fiber + 10grms Horn Material.**



**Figure 31 BOUNDARY CONDITIONS  
FORCE: 500N**



**Figure 33 shear stress of Kevlar fiber + 10grms Horn powder Material**

**ANALYSIS ON ANSYS  
STRUCTURAL STATIC ANALYSIS:**

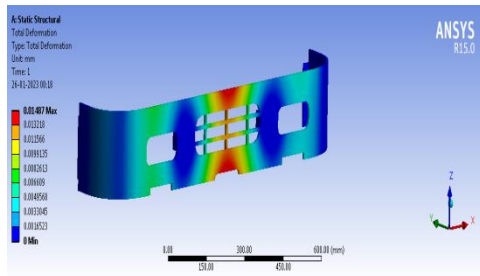


Figure 34 Total deformations of Kevlar fiber + 10grms Horn powder Material

STATIC ANALYSIS OF ALB390 MATERIAL:

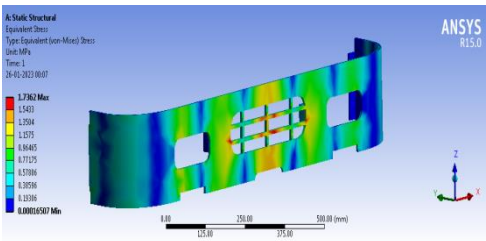


Figure 35 Von-misses stress of ALB390 Material

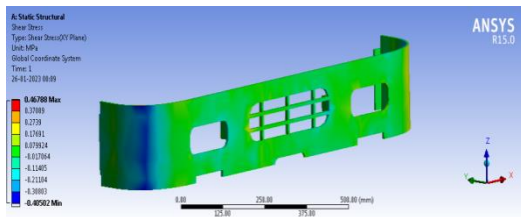


Figure 36 Shear stress of ALB390 Material

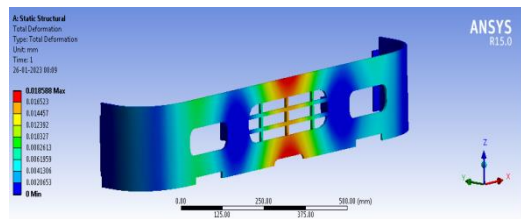


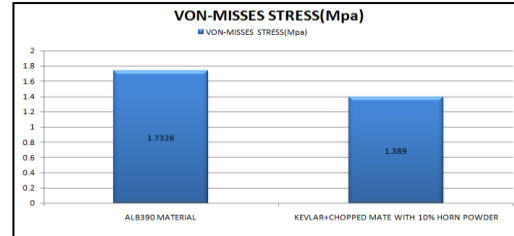
Figure 37 Total deformations of ALB390 Material

GRAPHS:

VON-MISSES STRESS GRAPH:

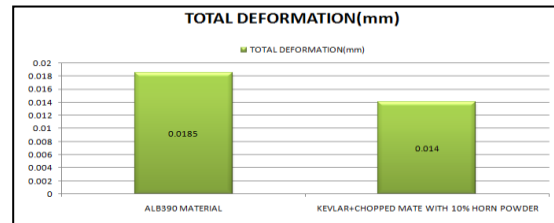
The below graph von-misses stresses of two different materials like KEVLAR+CHOPPED MATE WITH 10% HORN POWDER and ALB360 MATERIAL. Finally observed

KEVLAR+CHOPPED MATE WITH 10% HORN POWDER Have low von-misses stress 1.389Mpa compared to ALB360



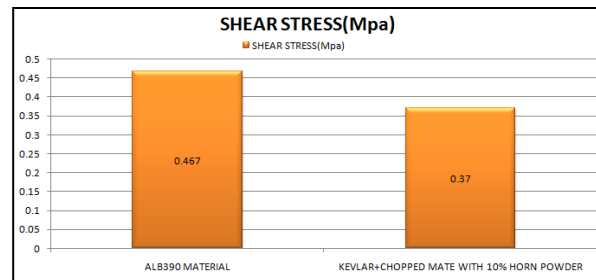
Graph 6 Von misses stresses of existing and proposed material

TOTAL DEFORMATION GRAPH:



Graph 7 Total deformation for existing and proposed material

SHEAR STRESS GRAPH:



Graph 8 Strain for existing and proposed material.

CONCLUSION

1) Compared to samples of composite epoxy reinforced with basalt fibres and the other hybrids, the tensile stress in the hybrid composite epoxy reinforced with chopped mate with Horn powder resin and Kevlar fibres has increased significantly. Since basalt and chopped matt fibres are randomly distributed, Kevlar fibres with chopped matt are continuous and have a lower tensile strength and modulus.

2) The hybrid composite (epoxy reinforced with chopped mate and Kevlar fibres) has a better value in terms of flexural strength than the sample of epoxy chopped Horn powder fibre composite because Kevlar fibres adhere to epoxy more readily than basalt fibres do.

3) The hardness test results indicate that the hybrid composite (epoxy chopped Kevlar fibres composite) has a lower Brinell number composite value. Hardness often reduces as a material's elasticity rises.

4) The presence of Kevlar fibres in the hybrid composite, which have a higher strength and impact resistance, results in a sample of the composite reinforced with chopped mate and Horn powder Kevlar fibres having a higher impact strength value than the sample of composite reinforced with basalt and chopped mate fibres.

5) Better stress, strain, and deformation values for chopped matt with Kevlar among the available Kevlar fibre are also shown by modelling and static analysis. After conducting several tests and taking into account static analysis, we have ultimately determined that combining Kevlar with chopped matt and horn powder is preferable to reaming six compositions.

6) The car bumper was designed using CATIA software, and its analysis was conducted using Ansys software using two materials. 10% HORN POWDER KEVLAR+CHOPPED MATE INCLUDING ALB360 MATERIAL Ultimately, the ideal material is KEVLAR+CHOPPED MATE WITH 10% HORN POWDER due to its low von-misses stress, deformation, and shear stress.

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